Photo History of the Simplesat Experiment

Dave Skillman, P.I.

Initial GPS tests atop Bldg 6

- demonstrated 1 degree accuracy
- typical rate noise 200 arcsec/sec





Initial construction of telescope tube

- vertical rails framed by octagonal rings

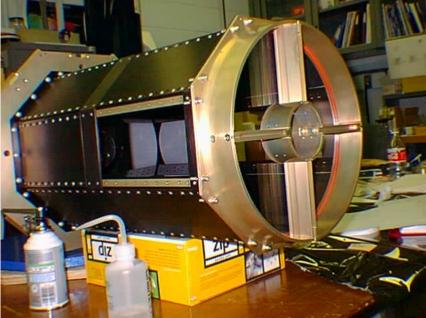




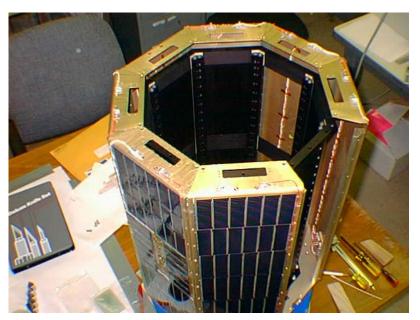
Inner frame panels attached

- baseplate/marmon (left)
- secondary mirror support (right)



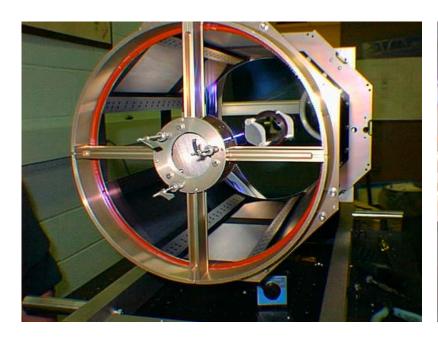


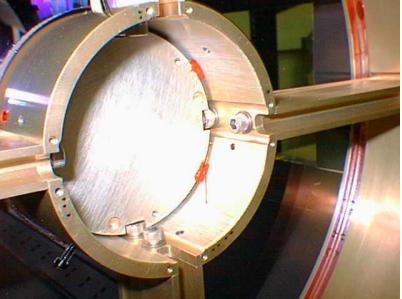
Fit check of solar array panels, primary mirror preps





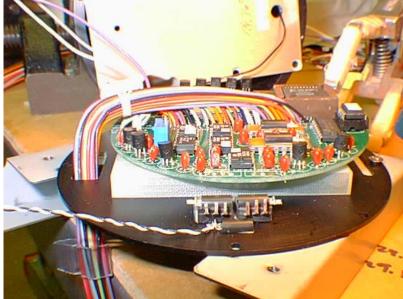
Align secondary mirror and tack into place - corrector plate held by RTV on outer edge





Camera ruggedize by remounting circuit board





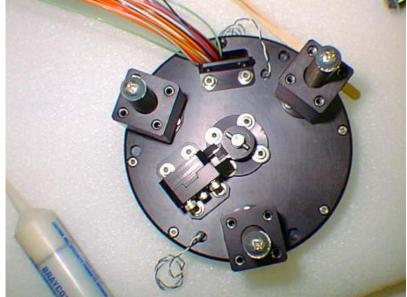
Circuit board epoxied to G-10 block (mechanical/thermal)





Reworked camera is mounted to focus table





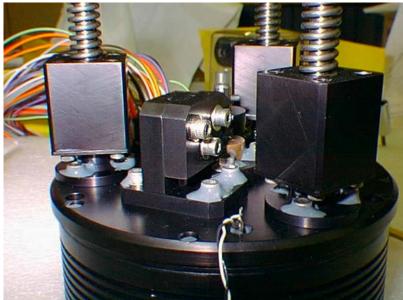
Focus table lifting thread, limit block, preload springs and motor



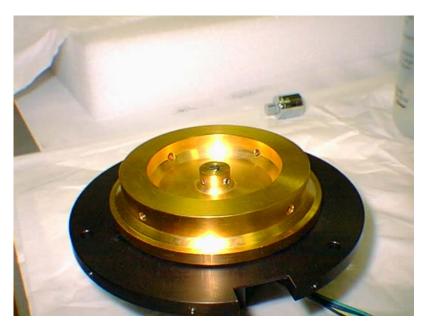


Focus mechanism staking with flexible epoxy





Optical (X) axis reaction wheel with stepper drive





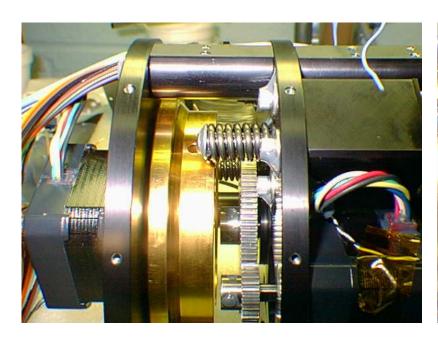
Focus mechanism assembly

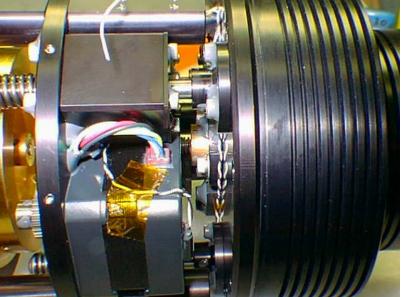




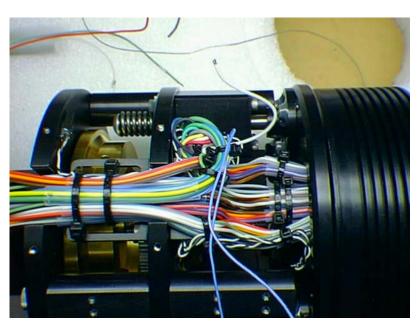
Focus mechanism details

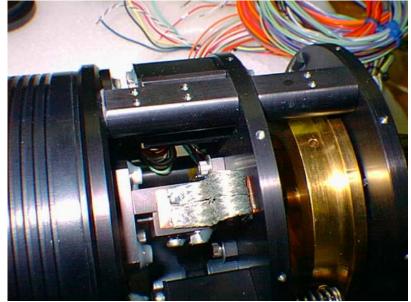
- reaction wheel and focus drive gears (left)
- focus motor, table, camera (right)



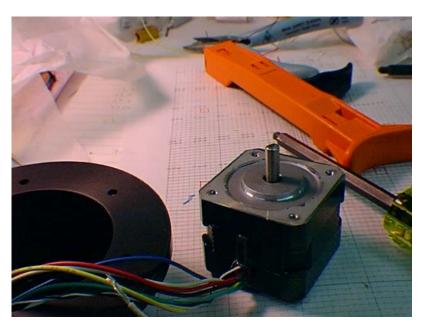


Camera/focus wiring and thermal shunt



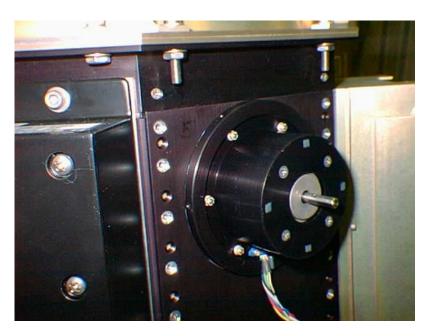


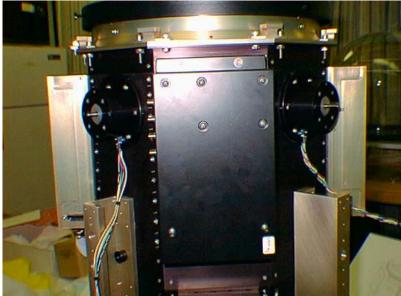
Body reaction wheel stepper motor and mount



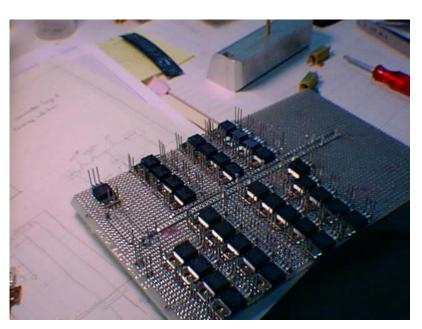


Y and Z axis reaction wheel motor mountings





Power distribution box and electronics board





Final power distribution circuit board (top/bottom)





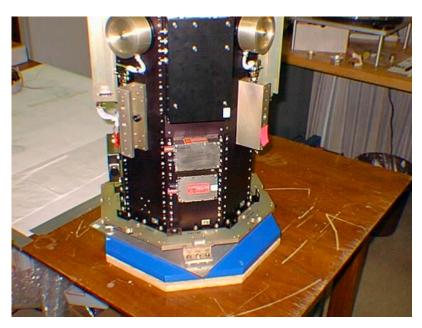
Separation switch mounting plate

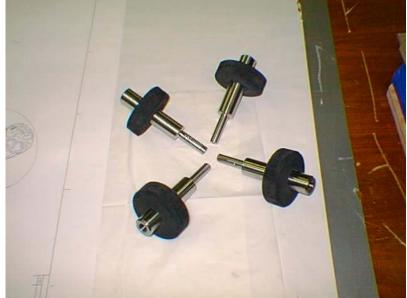
- each arm adjustable to mate with push plate





Array spacer blocks and support pins





Spacecraft handling box and general purpose fixture



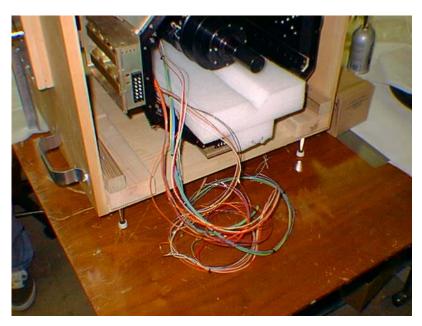


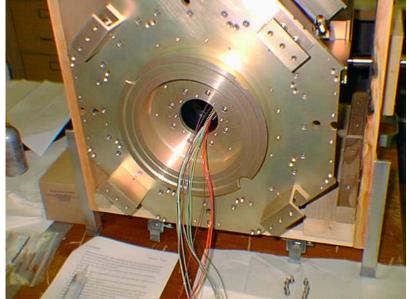
Box rotated right to attach lower end closure



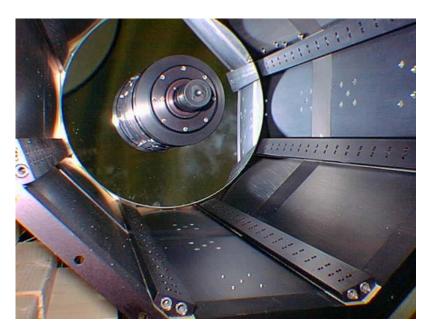


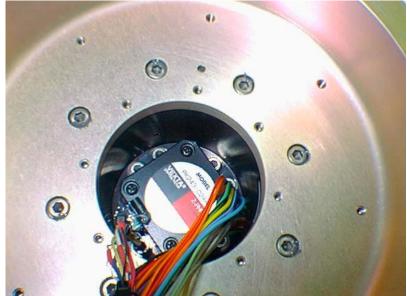
Box used to support spacecraft during camera installation



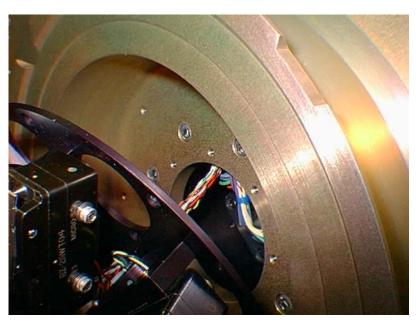


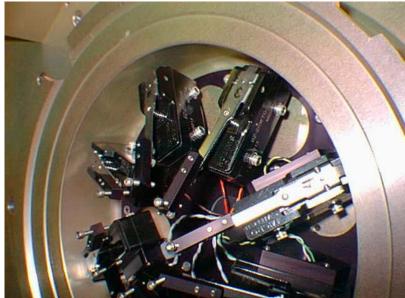
Camera installed, reaction wheel motor visible through baseplate



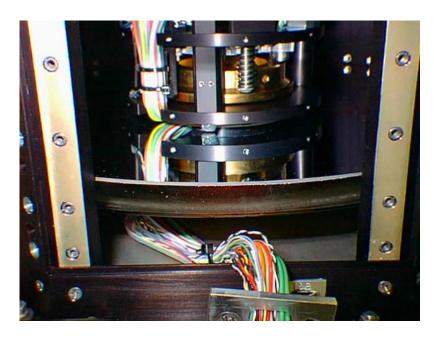


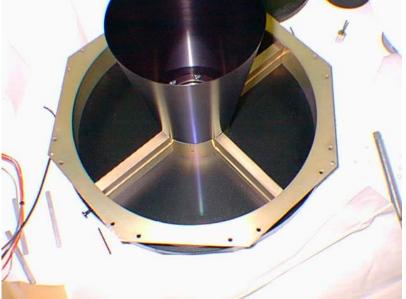
Fit check of separation switches





Camera/focus-mech and secondary baffle installation





Star-imaging test through roof hatch with sidereal drive



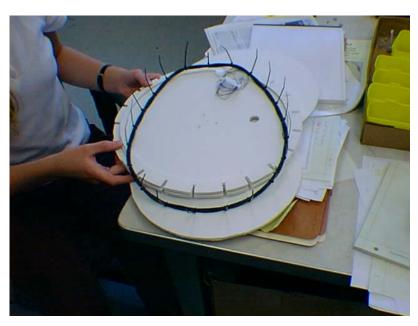


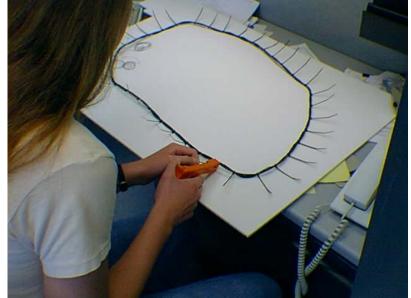
Typical temperature sensor mounting on electronics



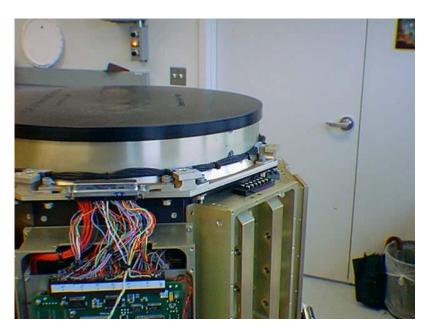


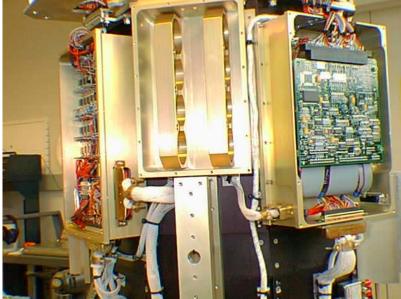
Summer student winding torque coils



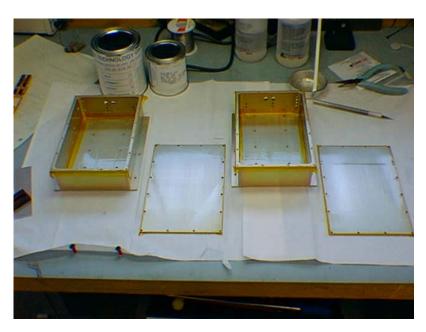


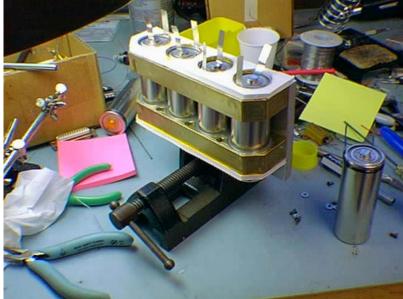
X-axis torque coil mounting (left) Battery box fit check (right)



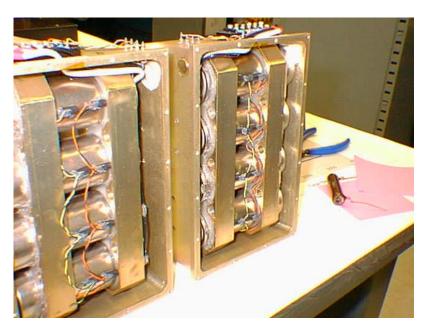


Battery box coatings and cell mounting



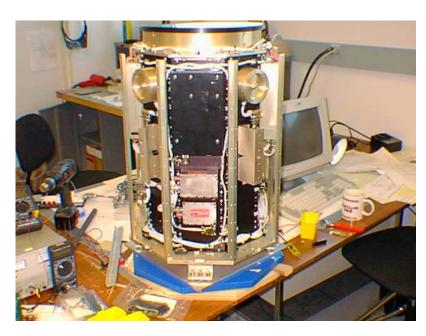


Finished battery boxes and KOH absorber installation



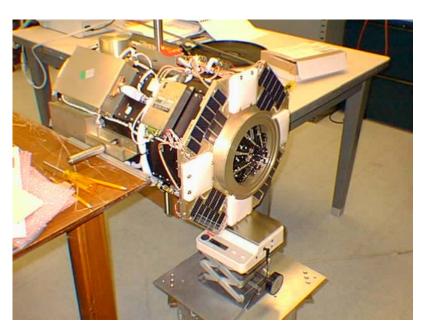


Getting ready for first balance test (no arrays)



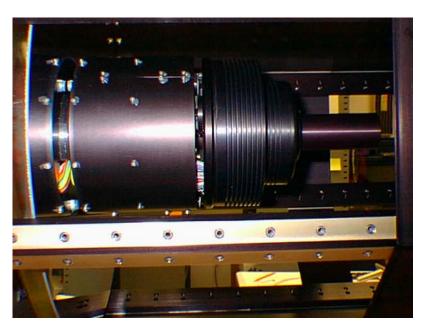


Balance tests locate center-of-mass





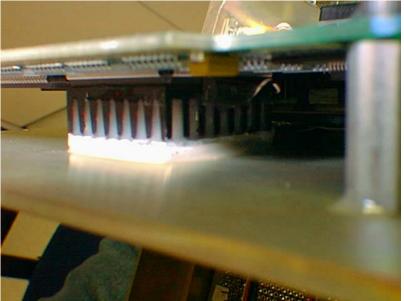
Camera/Focus-mech shrouds and baffles installed





CPU thermal mounting





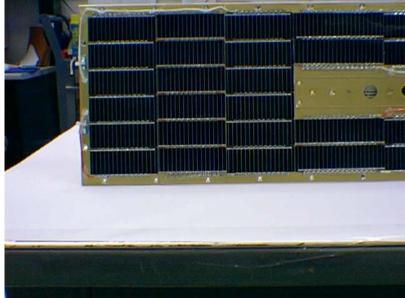
Removal of GPS lithium battery and thermal vacuum test of CPU boxes



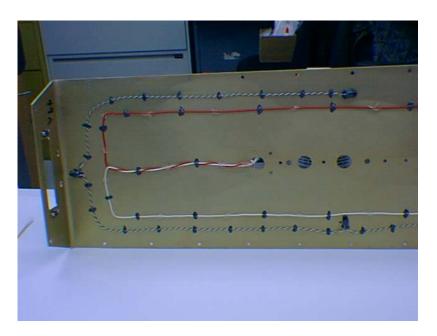


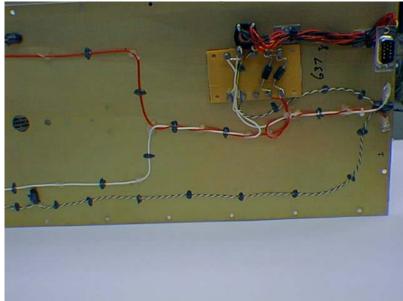
Summer student mounts thermal sensors on array panels





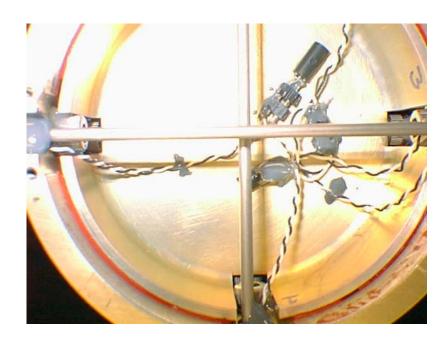
Arrays are back-wired to cancel magnetic fields





Power distribution box RF dc/dc convertors (left) and Y/Z hysteresis rods (right)





Harness tie-down details

- note reaction wheel cover (right)





Pre-vib checkout with CPU peripherals attached



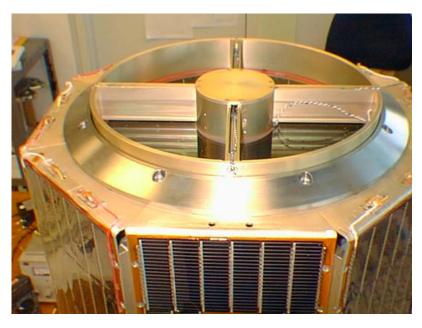


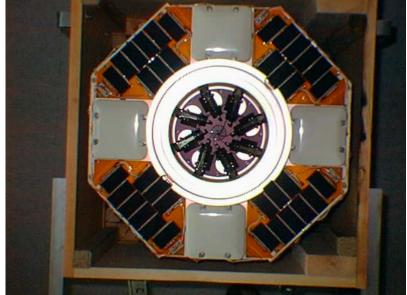
Pre-vib CG check (inertia moments by model)



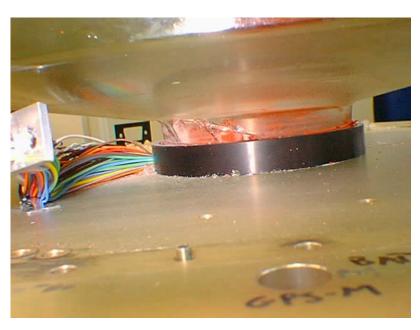


Pre-vib packing into carrying box





Primary mirror bond failure during vib test



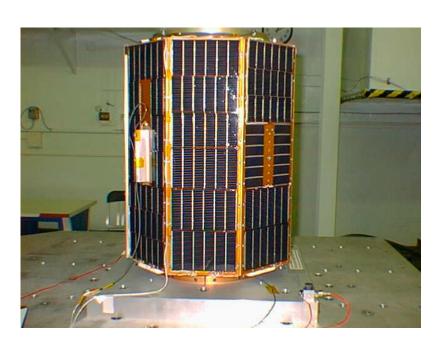


New primary and rebuilt support structure



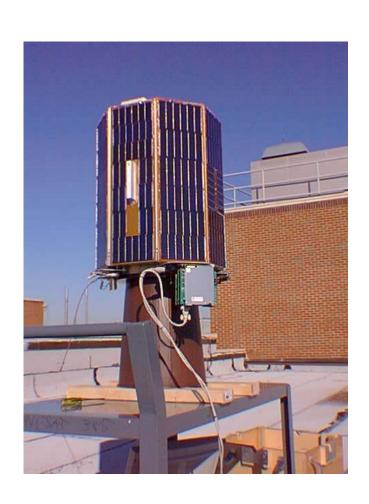


Final vib test successful





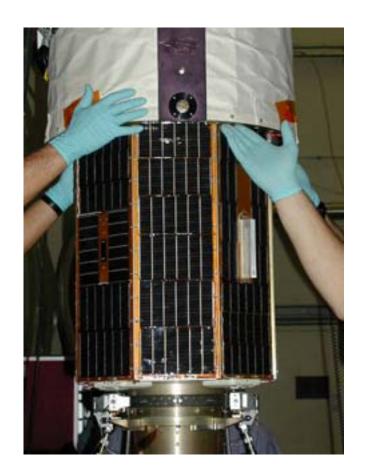
Solar array checkout and GPS self-survey



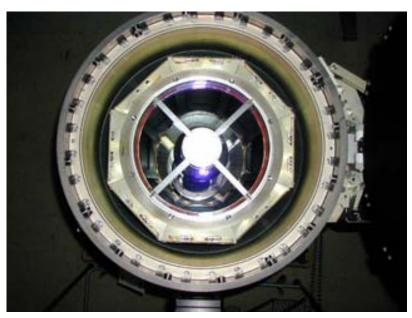


Simplesat on ejection system, canister installed





Simplesat ready-to-ship, installed in Discovery (STS-105)





Ground station tower and RF gear





Ground station tower and Yagi antenna atop Bldg33





Ejected into orbit 20 Aug 2001 over Lake Michigan

